FINITE ELEMENT ANALYSIS OF SELF-PIERCE RIVETING IN MAGNESIUM ALLOYS SHEETS

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ABSTRACT

Conventional fusion joining methods such as resistance spot welding have been demonstrated to not be effective for magnesium alloys. Therefore, self-pierce riveting (SPR) has been presented as an attractive joining technique for these lightweight metals. However, SPR must be performed at elevated temperatures because of the low ductility of magnesium alloys at room temperature. Even though the SPR joining process has been established on magnesium alloys, this joining process is not optimized. As such, this study establishes the first attempt at simulating the SPR of magnesium alloys through the use of the finite element method. An internal state variable (ISV) plasticity and damage material model was employed with results in good agreement to experimental data. The results of this study show that the ISV material model is ideally suited for modeling the SPR process in magnesium alloys.
DEDICATION

This thesis is dedicated to everyone who helped me during this period. In particular, my family and close friends, who always cheered and supported me.
LIST OF ABBREVIATIONS AND SYMBOLS

SPR  Self-pierce riveting
ISV  Internal state variable
RSW  Resistance spot welding
DRX  Dynamic recrystallization
FEA  Finite element analysis
BCJ  Model created by Bammann, Chiesa and Johnson

F  Deformation gradient
F_e  Deformation gradient, elastic component
F_p  Deformation gradient, plastic component
F_d  Deformation gradient, damage component
F_{th}  Deformation gradient, thermal component
L_e  Elastic velocity gradient
L_d  Velocity gradient associated with damage
L_p  Plastic velocity gradient
L_{th}  Velocity gradient, thermal component
D_e  Symmetric part of the elastic velocity gradient
D_p  Symmetric part of the plastic velocity gradient
D_d  Symmetric part of the velocity gradient associated with damage
D_{th}  Symmetric part of the velocity gradient, thermal component (thermal stretching)
$\phi$  Damage or porosity

$W_e$  Skew symmetric part of the elastic velocity gradient

$W_p$  Skew symmetric part of the plastic velocity gradient

$W_d$  Skew symmetric part of the velocity gradient associated with damage

$W_{th}$  Skew symmetric part of the velocity gradient, thermal component

$SSD$  Statistically stored dislocations

$\alpha$  Kinematic hardening internal state variable

$\kappa$  Isotropic hardening internal state variable

$T$  Temperature

$\lambda$  Lamè constant

$\mu$  Shear modulus

$\dot{\sigma}$  Conveceted Cauchy stress

$A$  Expansion coefficient

$GND$  Geometrically necessary dislocations

$C_v$  Heat capacity

$\rho$  Density of the material

$Y(T)$  Determines the rate-independent limit of yield stress

$f(T)$  Determines the strain rate at which the yield behavior exhibits a transition from being essentially rate-independent to being rate-dependent

$V(T)$  Determines the magnitude of the rate-dependence

$r_s(T)$  Kinematic static recovery parameter

$R_s(T)$  Isotropic static recovery parameter

$r_d(T)$  Kinematic dynamic recovery parameter

$R_d(T)$  Isotropic dynamic recovery parameter
**h(T)**  Kinematic hardening parameter

**H(T)**  Isotropic hardening parameter

**C_1**  Constant term in V(T) Arrhenius-type equation which determines the magnitude of rate dependence on yielding (units=MPa)

**C_2**  Temperature dependent activation term in V(T) Arrhenius-type equation (units=Kelvin)

**C_3**  Constant term in Y(T) Arrhenius-type equation which is the rate independent yield stress (units=MPa)

**C_4**  Temperature dependent activation term in Y(T) Arrhenius-type equation (units=Kelvin)

**C_5**  Constant term in f(T) Arrhenius-type equation which determines the transition strain rate from rate independent to dependent yield (units=1/sec)

**C_6**  Temperature dependent activation term in f(T) Arrhenius-type equation (units=Kelvin)

**C_7**  Constant term in \(r_d(T)\) equation which describes the kinematic dynamic recovery (units=1/MPa)

**C_8**  Temperature dependent activation term in \(r_d(T)\) equation (units=Kelvin)

**C_9**  Constant term in \(h(T)\) equation which describes the kinematic anisotropic hardening modulus (units=MPa)

**C_{10}**  Temperature dependent activation term in \(h(T)\) equation (units=Kelvin)

**C_{11}**  Constant term in \(r_s(T)\) equation which describes the kinematic static recovery (units=1/(MPa sec))

**C_{12}**  Temperature dependent activation term in \(r_s(T)\) equation (units=Kelvin)

**C_{13}**  Constant term in \(R_d(T)\) equation which describes the isotropic dynamic recovery (units=1/MPa)

**C_{14}**  Temperature dependent activation term in \(R_d(T)\) equation (units=Kelvin)

**C_{15}**  Constant term in \(H(T)\) equation which describes the isotropic hardening modulus (units=MPa)

**C_{16}**  Temperature dependent activation term in \(H(T)\) equation (units=Kelvin)

**C_{17}**  Constant term in \(R_s(T)\) equation which describes the isotropic static recovery (units=1/(MPa sec))
\( C_{18} \) \hspace{1em} \text{Temperature dependent activation term in } R_s(T) \text{ equation (units=Kelvin)}

\( CF \) \hspace{1em} \text{Coefficient of friction}
ACKNOWLEDGMENTS

I would like to thank the many colleagues, friends, and faculty members who have helped me with this research project. I am very grateful to Dr. James Brian Jordon, my advisor and chairman of this thesis, for sharing his research expertise and motivating me. I would also like to thank all of my committee members, Dr. Yuebin Guo for his support, and Dr. Mark E. Barkey for his valuable contribution regarding the finite element analysis (FEA). I am very thankful to Dr. Wang at GM for sharing experimental data, and to Dr. Douglas J. Bammann for his priceless contribution concerning the FEA and the material model used. Lastly, I am indebted to Department of Mechanical Engineering at The University of Alabama for the support of this project.

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1. INTRODUCTION

The need to decrease fuel consumption and greenhouse gas emissions have led the automotive industry to seek materials with enhanced strength-to-weight ratios [1]. Suitable materials such as aluminum and magnesium alloys are being employed in body-in-white construction. In fact, in certain cases, the use of lightweight metals like magnesium alloys can save significant weight while maintaining acceptable structural performance. Nonetheless, traditional joining techniques like resistance spot welding (RSW), traditionally applied to join steel metal, present enormous difficulties in joining magnesium alloys [2,3]. These difficulties are mostly due to the high electrical conductivity and low heat generation of magnesium alloys which in turn requires high welding currents [4] leading to poor welding characteristics. While many joining technologies exist, self-piercing riveting (SPR) is an appealing joining processes for magnesium alloys sheets [5].

The SPR process can be illustrated in four steps, as shown in Fig. 1: (1) the blank holder presses the two sheets against the die; (2) the punch drives a semi-tubular rivet into the upper sheet; (3) the material of the bottom plate flows into the die and the rivet shank begins to flare outward, forming a mechanical inter-lock between the two sheets; (4) and lastly, the punch is released [6].
It has been demonstrated that the SPR process produces joints of comparable static strength and superior fatigue performance to RSW and also produces promising results in peel and shear testing [7]. Although the SPR has many benefits, it does present some disadvantages including the following: both sides of the sheet must be accessible to apply the SPR technique; it is unsuitable for brittle substrates; prominences and indents related to the forming process may not be aesthetically acceptable; and relatively high forces are required for the forming process. As such, it is the difficulty of applying the SPR process to brittle materials that represents an obstacle in joining magnesium alloy sheets. In fact, employing the SPR process on AZ31 magnesium alloy at low or intermediate temperatures (i.e. lower than 150°C) results in cracks in the deformed material that typically originate in the shear zones where the strain localization arises due to limited dynamic recrystallization (DRX). The high rate of deformation in SPR does not permit sufficient time for DRX to occur during the deformation. Since DRX is improved at elevated temperatures, cracking of the joint can be prevented by preheating the AZ31 magnesium
alloy prior to SPR. In fact, experiments have shown that riveting AZ31 magnesium alloy at elevated temperatures largely removed the defects from the joint at temperatures of 200°C or higher [5,8]. In addition, it has been demonstrated that the joint strength increased when the preheat temperature was raised from ambient to 200°C. This increase in strength is due to the reduction in joint discrepancies and an increase in mechanical interlock between the rivet and the work pieces [5].

In regards to modeling the SPR process, several numerical studies on aluminum alloys have been conducted. One of the first to model the SPR process, Porcaro et al. [6] demonstrated good comparison between experimental results and finite element analysis (FEA) by using LS-DYNA 2D axisymmetric model while employing r-adaptivity re-meshing techniques and a geometrical failure criterion based on the change in sheet thickness. Casalino et al. [9] also showed good agreement between experimental and finite element results, but used a failure criterion based on effective plastic strain rather then a sheet thinning technique. Several other numerical studies of the SPR process have been conducted with similar results [10,11]. Regarding the springback analysis, the common approach is to perform the piercing step in explicit and perform the springback using implicit [6,9–11].

While recent FEA of SPR on aluminum alloy sheets has demonstrated the effectiveness of these modeling techniques, to the best of the authors’ knowledge, there is no study published work regarding FEA of SPR in magnesium alloys. More importantly, previous studies [6,9–11] have not considered strain rate, temperature, stress-state dependencies, or heat generation due to plastic deformation, as these effects are not easily captured by simple piece-wise linear material models. As such, the purpose of this paper is to, for the first time, use a large scale plasticity and
damage model based on internal state variables (ISV) to capture the complex physics occurring during the SPR process and compare to experimental results.
2. NUMERICAL TECHNIQUE

The material model employed in this study is the internal state variable (ISV) plasticity and damage based upon ductile void growth by Bammann [12], Bammann & Aifantis [13] and Bammann et al. [14] (also called the BCJ model). This model was later modified to capture damage arising from void nucleation, and coalescence, in addition to the void growth. [14-15], and later modified to incorporate more accurate physics [16]. It has been used to predict the behavior of metals under different loading conditions [17] on many different industrial and manufacturing applications (including forging, metal cutting, heat treatment, resistance welding) [18]. The BCJ model is a physically based plasticity and damage constitutive model that involves microstructural content and is consistent with continuum level kinematics, kinetics, and thermodynamics [19]. This model was originally developed for problems with strain rates smaller than the shock regime, involving large strains and or damage. Generally, whenever either or both of these mechanisms are present, localizations occur, driven by gradients in strain/damage and inherently coupled to the softening resulting from the self-heating. For this reason, the BCJ model can be applied to simulate a self-pierce riveting process even at high temperatures and strain rates due to the kinematic and isotropic hardening and recovery components.

The kinematics of the BCJ model are based upon the multiplicative decomposition of the deformation gradient into elastic, damage and plastic parts.

\[ F = F_e F_d F_p \]  \hspace{1cm} (1)
As a result, other kinematic quantities naturally follow and are defined with respect to any of the possible configurations via standard push–forward, pull–back of any of the configurations operations. For example, the velocity gradient naturally decomposes in the current configuration into the following additive parts.

$$L = \dot{F} F^{-1} = \dot{F}_e F_e^{-1} + F_e \dot{F}_e F_e^{-1} + F_e F_e \dot{F}_e F_e^{-1} + F_e F_e F_e \dot{F}_e F_e^{-1} + F_e F_e F_e F_e \dot{F}_e F_e^{-1}$$  \hspace{1cm} (2)

Where, $L_e = \dot{F}_e F_e^{-1}$ is the elastic the velocity gradient defined with respect to the current configuration, $L_d = \dot{F}_d F_d^{-1}$ is the velocity gradient associated with damage and is defined with respect to the configuration $F_d$, while $L_p = \dot{F}_p F_p^{-1}$ is the plastic velocity gradient residing in the configuration defined with respect to $F_p$. The velocity gradient can obviously be pulled back to any of the configurations, as can any of the finite deformation strain measures. However, it is noteworthy that the total strain or velocity gradient is simply the sum of the elastic damage and plastic parts, each defined by the appropriate pull–back operation. The BC J model is defined with respect to the configuration that is described by $F_p$, the configuration reached by unloading through the elastic and damage configurations. The Helmholtz free energy was written as a function of the elastic strain and the ISV all described with respect to this configuration. The assumption that the free energy is a quadratic function of the elastic strain, results in a Hooke’s law the expression, which after taking the material derivative and pushing forward to the current configuration yields an expression (in rate form) for linear elasticity in the current configuration.

$$\dot{\sigma} = \lambda (1 - \phi) TrD_e + 2\mu (1 - \phi) D_e$$  \hspace{1cm} (3)

where $D_e$ is the symmetric par of the elastic velocity gradient and $\phi$ is the porosity or damage.
Recall, in each velocity gradient $L$ may be decomposed into a symmetric part, $D$ and a skew symmetric part $W$.

\[
L = D + W \quad (4)
\]

\[
L_e = D_e + W_e \quad (5)
\]

\[
L_p = D_p + W_p \quad (6)
\]

\[
L_d = D_d + W_d \quad (7)
\]

\[
L_{th} = D_{th} + W_{th} \quad (8)
\]

Since the damage and thermal parts of the deformation gradient are assumed to be isotropic, there is no skew symmetric part for either. So,

\[
D = D_e + D_p + D_d + D_{th} \quad (9)
\]

\[
W = W_e + W_p \quad (10)
\]

but since the plastic part is neglected in the BCJ model,

\[
W_p = 0 \Rightarrow W_e = W \quad (11)
\]

therefore, to complete the model, constitutive equations are necessary for $D_p, D_d$ and $D_{th}$. These flow rules are given in Eq. 12, where $\sigma'$ is the deviatoric Cauchy stress and $\alpha'$ is the tensor variable. $D^{th}$ is the thermal stretching (Eq. 13) assuming isotropic thermal expansion where $A$ is the expansion coefficient, $I$ is an unit tensor and $T$ is the temperature, and assuming an isotropic volumetric rate of change due to void growth in terms of the porosity or damage $\phi$ in equation (14).
The kinematic hardening internal state variable \( \alpha \) and the isotropic hardening internal state variable \( \kappa \) are given in Eqs. 15 and 16, respectively. The evolution of the internal state variables \( \alpha \) and \( \kappa \) is defined in a hardening minus recovery format. The isotropic hardening variable \( \kappa \), is motivated by densities of statistically stored dislocations (SSD) and are generally created and stored within grains. The kinematic hardening variable \( \alpha \), is motivated by geometrically necessary dislocations (GND) or compatibility dislocations that are created to restore compatibility at boundaries between parts of the crystal oriented differently with respect to each other, such as grain boundaries. Dislocations of the same sign piled up against a boundary are also represented as GNDs and their motion under load reversal gives rise to an apparent softening or a Bauschinger effect.

\[
\dot{\alpha} = h(T)\dot{D}^p - \left[ r_d(T)\|D^p\| + r_v(T)\|\alpha\|\right] \alpha
\]

\[
\dot{\kappa} = H(T)\dot{D}^p - \left[ r_d(T)\|D^p\| + r_v(T)\|\kappa\|\right] \kappa^2
\]

Temperature change due to deformation is taken into account in Eq. 17, where \( \rho \) is the density of the material, and \( C_v \) is the heat capacity. In this equation adiabatic temperature change is assumed, where 90% of the plastic work is dissipated as heat. The evolution of the damage (void growth) parameter \( \phi \), is expressed in Eq. 18 and 19.
Equations (18) and (19), describing ductile void growth were developed by Cocks and Ashby [20, 21]. Notice the strong dependence on the stress triaxiality. At high pressures equation (19) and equation (14) and the spherical part of Hooke’s law (pressure, where the bulk modulus is degraded by the damage) describe a competitive process. As increased pressure drives the growth of porosity, the increased porosity slows the evolution of the pressure. This leads to a natural saturation of the porosity. Equation (14) is often neglected at low to moderate pressures, as the degradation of the elastic moduli by the damage is sufficient to predict the evolution of the porosity. Also, equation (14) causes a strong numerical stiffness in the solution of the system of ordinary differential equations and results in more enhanced solution techniques and greatly increases the computational time.

The model parameters $V; Y; H; R_d; R_s; h; r_d; r_s$, are functions of temperature $(T)$, stress state and strain rate and are correlated to experiments. The rate-independent limit of yield stress is set by $Y(T)$. The $f(T)$ term determines the strain rate at which the yield behavior exhibits a transition from being essentially rate-independent to being rate-dependent and the $V(T)$ term determines the magnitude of the rate-dependence. The $H(T)$ and $h(T)$ terms are the hardening moduli, $r_s(T)$ and $R_s(T)$ are scalar functions describing the diffusion controlled ‘static’ or ‘thermal’ recovery, and $r_d(T)$ and $R_d(T)$ are the functions describing dynamic recovery. The

\[
T = \frac{0.9}{\rho C_v} (\sigma \mathbf{D}^\rho) \quad (17)
\]

\[
\dot{\phi} = \beta \left[ \frac{1}{(1-\phi)^\nu} - (1-\phi) \right]^{D^\nu} \quad (18)
\]

\[
\beta = \sinh \left[ \frac{2(2N-1)\rho}{(2N-1)\sigma} \right] \quad (19)
\]
deviatoric inelastic flow rule, Eq. 15, captures the effect of creep and plasticity in terms of kinematic and isotropic hardening, temperature, and strain rate.

In this study, the numerical results are compared to SPR produced by Wang et al. [5]. In this study, 200°C and 110 mm/sec were experimentally found to be the optimal temperature and riveting velocity, respectively, for producing defect free joints. As such, these parameters were chosen as a baseline for SPR simulations. The model constants $C_1$ to $C_{18}$ (refer to Appendix A for equation $C_1$–$C_{18}$) were calibrated for the stress-strain curves for AZ31 at various strain-rates. The calibrated model was verified by comparing experimental literature data with simulations of a tensile test of a single element in LS-Dyna with dimensions 1x1x1 mm, and a 0.1x0.1 mm element 2D axisymmetric shell for high strain rates applied in the explicit method. Figure 2 shows the comparison of the ISV plasticity model to experimental tensile stress-strain results for room temperature quasi-static [22] quasi-static at 370°C [23], and 300°C at strain rate of 1000/s [22]. All stress-strain data used here was obtained from axial loading parallel to the rolling direction of the sheet [22, 23]. As shown in Fig. 2, the ISV model provided good correlation to the plastic flow stress of the AZ31 magnesium alloy.

In this study, all of the SPR simulations were conducted using axisymmetric 2D analysis in LS-dyna. The ISV material model described earlier is included in LS-dyna [24] and is listed as *MAT_BAMMAN (material mode type 51) and *MAT_BAMMAN_DAMAGE (material model type 52). To be consistent with the experiments by Wang et al., [5], a 0.35% carbon steel with 280HV was used by to represent the rivet material. An elastic-plastic material (*MAT_PIECEWISE_LINEAR_PLASTICITY), type 24 in LS-Dyna [24] was used for the rivet. Finally, the deformations of the pad, die and punch exhibit very low deflection compared to the
rivet and sheets, and thus, were modeled as rigid bodies (*MAT_RIGID, type 20 in LS-Dyna [24]).

Typically, in solid mechanics, the Lagrangian formulation with uniform fixed element meshes is mainly used in FEA [6]. However, the drawback of this method is that, in large deformation simulations, elements may become very distorted, compromising the accuracy of the model. In order to avoid complications due to mesh disturbances and to capture the effect of the rivet piercing the sheet, the kill element technique was applied after damage of 0.99 is reached. In LS-Dyna 2D explicit, damage does not reach 0.99 due to numerical issues [25], and thus a erosion criterion is included in *MAT_BAMMAN_DAMAGE. As the damage value reaches 0.99, the damage stiffness and yield of the element is 1% of the original values at the initial state. In fact, the strain-to-failure criterion must be large enough in order to eliminate elements that reached damage of 0.99 and sufficiently small to avoid high element distortion and numerical complications.

![Figure 2](image.png)

**Figure 2.** Comparison between the experimental and the BCJ model correlation results of a tensile test.
In the current study, the Lagrangian method with an optimized mesh was used to decrease the time of the calculation. Since the SPR process can be approximated as axisymmetric problem, a four node 2D axisymmetric element was used, with four Gauss integration points and a stiffened based hourglass control (strain co-rotational stiffness form). Contact was modeled by applying *2D_AUTOMATIC-SURFACE_TO_SURFACE penalty formulation contained in LS-Dyna. Since the SPR analysis is dynamic, an explicit calculation mode was performed to model the piercing process. After the SPR process is completed, a springback and cooling step analysis was performed. As the material model used for magnesium cannot be applied in shell elements utilizing an implicit calculation, these two post-process calculations were performed using an explicit method, activated by *DYNAMIC_RELAXATION command in LS-Dyna.

As described elsewhere [5], the AZ31 magnesium alloy sheets were 2 mm in thickness. The geometric characteristics of the die (type DZ1002050H1R1.5) and rivet (K50642AH00) are show in Fig. 3 and the dimensional values are listed in Table 1. The punch has a radius of 4 mm and a height of 0.8 mm. The pad has an outer radius of 9.25 mm and an inner radius of 5.45 mm, and the bottom corners are rounded with a radius of 0.2 mm. Regarding the SPR process, the clamping speed was 5 mm/second, the force applied on the holder to keep the sheets clamped was 600 N, the riveting speed was 110 mm/s and the riveting displacement was 6.25 mm downwards. Since the FEA model is axisymmetric, the clamped force is 600 N per unit radian.
Table 1. Dimensions of the self-pierce riveting die and rivet

<table>
<thead>
<tr>
<th>Part</th>
<th>D (mm)</th>
<th>h (mm)</th>
<th>t (mm)</th>
<th>d (mm)</th>
<th>L (mm)</th>
</tr>
</thead>
<tbody>
<tr>
<td>Rivet</td>
<td>-</td>
<td>-</td>
<td>-</td>
<td>5</td>
<td>6</td>
</tr>
<tr>
<td>Die</td>
<td>10</td>
<td>1.5</td>
<td>0.25</td>
<td>-</td>
<td>-</td>
</tr>
</tbody>
</table>

The initial configuration of the finite element simulation of the SPR process is shown in Fig. 4. The size of the smallest element in the sheets and the rivet is 0.1 mm x 0.1 mm, as this was selected since Porcaro et al. [6] showed it gives a reasonable accuracy in the SPR simulations. In this configuration the entire model is composed by 5455 shell elements totaling 5903 nodes. Employing a finer mesh such as 0.05 mm x 0.05 mm in the large deformation area, resulted in 12535 elements and 13103 nodes, which increased the calculation time by about 15 times with relatively no change in accuracy compared to 0.1 mm x 0.1 mm mesh. Thus, the 0.1 mm x 0.1 mm mesh was used in this study.
The choice of the coefficient of friction (CF) in any FEA involving contact has a large influence in the total deformation. In fact, parameters such as temperature, pressure and velocity drastically influence the CF between steel and AZ31 magnesium alloy sheets [26] and previous SPR simulations have shown strong influence on the choice of the coefficient of friction and final deformation [6, 9]. However, in this study, an attempt was made to avoid letting the choice of CF to become a fitting parameter of the simulations. As such, we elected, possibly to the detriment of our simulation results, to use values that were closely related to experimental results. The choice of the friction was taken as a reference from Bolz & Tuve [27] and Ceretti et al. [28]. Thus, the coefficients used in this study are essentially based on physical parameters.

Figure 4. Initial configuration of the axisymmetric self-pierce riveting finite element model.
In addition, we note, that since the SPR process did not match exactly to the boundary conditions of the Bolz & Tuve [27] and Ceretti et al. [28], some variations were estimated taking into account change in temperature, pressure and velocity. Table 2 lists the CFs used in all of the simulations performed in this study. However, we acknowledge that future work in FEA of SPR should include CFs determined through experimental work with the same conditions as the SPR process. This could be done by conducting a pin-on-disk experiment using a pin made of steel and an AZ31 magnesium alloy disk at different temperatures, normal pressures and velocities similar to Ceretti E. et al. [28].

In this study, the following type of simulations using the ISV material model were performed at initial temperature of 200°C: (a) isothermal with a strain-to-failure based kill-element criterion correlated to experimental tensile tests (55% strain-to-failure [5]), (b) isothermal and strain-to-failure based kill-element criterion equal to 250% strain for top sheet, where this strain value was obtained via trial and error; (c) isothermal and strain-to-failure based kill-element criterion equal to 250% strain for both top and bottom sheets; d) adiabatic and strain-to-failure based kill-element criterion equal to 250% strain for on top sheet; e) adiabatic, kill-element criterion based damage reaching 0.99 on top sheet only; f) adiabatic, kill-element criterion based damage reaching 0.99 on both the top and bottom sheets. Springback analysis was performed on all of the simulations using a dynamic relaxation in explicit mode. We note that others have performed springback using an implicit approach. However, implicit implementation is not compatible with BCJ and shell elements in LS-dyna. Thus, we used a less common approach employing dynamic relaxation in an explicit code.
Table 2. Coefficients of friction (CF) values used for the simulations in this study

<table>
<thead>
<tr>
<th>Parts</th>
<th>Coefficient of Friction</th>
</tr>
</thead>
<tbody>
<tr>
<td></td>
<td>Static</td>
</tr>
<tr>
<td>Pad/Sheets</td>
<td>0.6</td>
</tr>
<tr>
<td>Die/Sheets</td>
<td>0.6</td>
</tr>
<tr>
<td>Rivet/Sheets</td>
<td>0.15</td>
</tr>
<tr>
<td>Sheet/Sheet</td>
<td>0.7</td>
</tr>
</tbody>
</table>

This explicit method has been used in other springback calculations in prior work, but not in any of the SPR simulation results shown in literature. Our springback simulations show that the overall shape after springback and cool down period does not significantly change the final shape of the joint.
3. RESULTS AND DISCUSSION

The application of SPR in Mg alloys is a complex large-scale rate and temperature dependent boundary value problem. As observed by Wang et al. [5] and elsewhere, the material during the riveting process experiences high strain rate, and physically observed temperature increase. As such, we selected the BCJ model, since the model was developed for these types of problems by capturing the physics and mechanics of large-scale plasticity, with the capability to significantly affect the strain to failure due to the inclusion of adiabatic heat generation. Figure 5 shows the an example of the SPR process at different time steps, showing effective stress (Fig. 5a) and effective strain (Fig. 5b) in both sheets. This image was obtained from the simulation with the following configuration: adiabatic, kill-element criterion based damage reaching 0.99 on both the top and bottom sheets. As the stresses in the rivet are much higher than the ones in the Mg sheets, the stress and strain shown are only for the Mg sheets. The comparison of the FEA results of this study are taken from the post-processing of the LS-dyna simulations. To make the comparison, springback and cooling are performed after the SPR process, so the outline of the FEA shape is superimposed on an image of the cross-section of the experimentally created joint, as shown in Fig. 5c [5]. To further aid in illustrating the stress and strain development in the SPR process, Video 1 and Video 2 show the complete time interval of the effective stress and effective strain evolution, respectively.
Figure 5. The self-pierce riveting process at different time steps: a) effective stress, b) effective strain, c) final shape of the FEA superimposed on the experimental results.

In order to illustrate the complexity of including damage in finite element analysis, visual evaluation of the FEA results with different configurations as described in the previous section, are compared to an image of the cross-section of the experimentally created joint [5]. Fig. 6a shows comparison of the isothermal SPR simulation using a simple tensile data obtained from Wang et al. [5] of AZ31 magnesium alloy at a temperature 200°C and strain rate of 4.1/s. As shown in Fig. 6a, the choice of a strain-to-failure of 55% produces a significantly unsatisfactory comparison to the experimental results. In fact, the kill-element criterion essentially eliminates the elements to soon, leading to a relatively brittle-like deformation not observed experimentally.
Figure 6b shows the visual evaluation of the SPR simulation with a strain-to-failure criterion chosen by trial-and-error (250%). In these simulations, the strain-to-failure was only applied to the top sheet and no damage was allowed to develop in the bottom sheet. As the strain-to-failure of 250% is significantly greater than the uniaxial tensile strain for this material at this temperature, it further illustrates the unsuitability of using uniaxial tensile strain to predict failure in a complex boundary value problem. For additional comparisons, the strain-to-failure criterion was applied to both the top and bottom sheets. As shown in Fig. 6c, using strain-to-failure criteria in the bottom sheet caused excessive softening of the bottom sheet leading to a reduction in the spreading of the rivet.

Figure 6d, e, and f, illustrate the results of allowing for heat generation during the deformation. Fig. 6d shows the results of using adiabatic and strain-to-failure of 250% on the top sheet only. The most obvious difference resulting from the inclusion of adiabatic heat generation in this simulation, was the gaps between the rivet and the sheets were largely eliminated. Fig. 6e shows the results of using adiabatic and the kill-element criterion based on damage reaching 0.99 on the top sheet only. In this simulation result, the gap between the rivet and the top sheet are also minimized. The main difference observed between the using as strain-to-failure and the kill-element criterion based on damage reaching 0.99 is the small amount of material underneath the rivet leg. In the strain-to-failure simulations (Fig. 6d), a majority of the elements of the top sheet just under the rivet leg are eliminated due to the uniform strain-to-failure criterion. However, in the simulations using the kill-element criterion based on damage reaching 0.99, a more representative amount of elements remain in the simulation similar to what is shown by the actual rivet.
Figure 6. Comparison between numerical analysis and experimental results [5] for the final shape of the SPR: a) isothermal and strain to failure based on uniaxial tensile results (55%); b) isothermal, strain to failure of 250% on top sheet; c) Isothermal and strain to failure of 250% on top and bottom sheets; d) adiabatic, strain to failure of 250% on top sheet; e) adiabatic and failure based on damage criterion on top sheet; f) adiabatic and failure based on damage criterion on top and bottom sheets. Experimental images adapted from [5].

In order to model the both the top and bottom sheet in a more physically representative approach, the bottom sheet was not shielded from damage, but allowed to evolve naturally. Figure 6f shows the results of using the kill-element criterion based on damage reaching 0.99 in both the top and bottom sheets. This simulation did not compare as well to the experimental results as of damage only in the top sheet. We further note, that the asymmetry of the AZ31 Mg alloy caused some difficulty in using the kill-element criterion based on damage reaching 0.99 in the bottom sheet. In fact, while not shown, simulation results showed that an excessive damage
rate was leading to premature softening not observed in the experimental results. As such, the damage growth rate was calibrated with a different value for the compression due to the asymmetry observed between tension and compression in magnesium alloy. The darker region of the bottom sheet in Fig. 4 was assigned a damage rate that allowed for larger elongation observed for Mg alloy under compression loading. The remainder of the bottom sheet was assigned the identical damage rate as the top. Using this anisotropic damage mapping, no elements of the bottom sheet reached a damage of 0.99.

To illustrate how temperature and stresses change during the SPR process, temperature and von Mises stress history plots taken from an element the top sheet near the vicinity of the rivet is presented in Fig. 7. The inclusion of heat generation in the plastic deformation is essential for accurately capturing the flow stress in the SPR simulation. In fact, Fig. 7a shows that for the element selected, for instance, a 160°C temperature increase was observed due to the large deformation. This large increase in temperature resulted in a decrease in the stress carried by this element of about 100MPa, as shown Fig. 7c. While these values are based on simulation results, we have illustrated the importance of including heat generation.

To evaluate the ability to predict cracks with ISV and damage model, and additional simulation was run (Fig. 8a) at an initial temperature of 25°C, with the adiabatic material model and damage applied on both top and bottom sheets, so the only parameter changed was the initial temperature of the sheets. The black elements in Fig. 8a were killed from the simulation and their location is shown in the deformed and final configuration of the joint. Figure 8b illustrates that the model predicted the onset of a crack in the same location compared to the experimental results shown in Fig. 8c.
Figure 7. History plot of one element from top sheet: a) element selected in green; b) temperature; c) von Mises Stress.
Figure 8. Comparison between numerical analysis and experimental results showing crack prediction at room temperature for SPR of Mg alloy: a) Location of the elements deleted on the final configuration of the joint during the simulation, b) elements deleted from [5]).

The use of the BCJ plasticity model appears to satisfactorily capture the complex nature of isotropic and kinematic hardening and recovery. In fact, we make the argument that the use of ISV models is likely a more computationally efficient and physically motivated approach to model the SPR process compared to other approaches. It is the combination of the hardening and recovery plastic flow rule coupled with heat generation that captures the deformation of SPR. Moreover, the BCJ model appears to adequately capture the large deformation without the need to remesh. This is a significant point, since remeshing algorithms can produce errors as large as 50%. More importantly, the advantage of not re-meshing is that that stress histories of
the elements remain in tact, resulting in the application of a stress-based damage criterion for material fracture.

While we acknowledge that others have produced good comparisons between simulations without an element deletion criterion, we have attempted to model the experimental results using a more physically accurate approach that models the damage model evolution. While the simulations results compared well to the experimental shape of the SPR with both strain-to-failure and damage model including adiabatic, we note that the simulation results with the damage model produced results that were less accurate in terms of final shape of the riveted join. The experiments required to adequately evaluate the triaxialty include notched tensile tests with a range of notch radii at various strain rates and temperatures. Unfortunately, we were unable to find data in literature on notched tensile tests at the required strain rates and temperature. This largely is a result that this type of experiment is very difficult to perform using a split Hopkinson bar test. Thus, we acknowledge that to truly calibrate the damage model, an extensive modeling and experimental effort is required, which is beyond the scope of this paper. Lastly we acknowledge that the BCJ model employed in this study does not explicitly address deformation due to twinning. However, at the ideal temperature (200°C) for performing SPR in Mg alloys, twinning is not activated and thus the flow rule of the BCJ is a valid assumption at elevated temperatures. Future modeling development of the BCJ model for HCP metals should include flow rules specifically for the deformation associated with twinning mechanisms.
4. SUMMARY AND CONCLUSIONS

Self-piecing riveting is a forming process in which a rivet, pressed by a punch, pierces the top sheet and flares into the bottom sheet, resulting in a mechanical interlock between the sheets. In the case of self-pierce riveting AZ31 magnesium sheets, the process must be conducted at elevated temperatures in order to produce a good joint due to the low ductility of this material at low temperatures. Numerical simulations of the SPR process were carried out using the commercial code LS-Dyna. The 2D axisymmetric model was composed of the rivet, two sheets, die, blank holder and punch. An explicit solution technique was applied and riveting through the top sheet was possible because of the use of a damage model and element deletion criterion.

For the first time, use of an internal state variable (ISV) plasticity and damage model was successfully applied to capture the complex deformation observed in the SPR. Good agreement was observed by visual comparison of the final geometry of the simulation and the experimental results. We further note that the good correlation of the deformation was achieved without re-meshing techniques used elsewhere in literature. Furthermore, the ISV model was shown to produce favorable predictions of cracks under room temperature riveting. Since the use of this ISV model is shown as a feasible method for modeling the SPR process, future simulation work could focus on different temperatures and punch velocities. In addition, this model could be used to check the feasibility of using aluminum rivets on magnesium sheets, resulting in a joint with better corrosive behavior. In addition, non-local methods should possibly be considered when damage or strain localization is present [29].
5. REFERENCES


APPENDIX A. ADDITIONAL ISV MODEL EQUATIONS

\[ V(T) = C_1 \exp\left(-\frac{C_2}{T}\right) \]  
\[ (16) \]

\[ Y(T) = C_3 \exp\left(\frac{C_4}{T}\right) \]  
\[ (17) \]

\[ f(T) = C_5 \exp\left(-\frac{C_6}{T}\right) \]  
\[ (18) \]

\[ r_d(T) = C_7 \exp\left(-\frac{C_8}{T}\right) \]  
\[ (19) \]

\[ h(T) = C_9 \exp\left(\frac{C_{10}}{T}\right) \]  
\[ (20) \]

\[ r_s(T) = C_{11} \exp\left(-\frac{C_{12}}{T}\right) \]  
\[ (21) \]

\[ R_d(T) = C_{13} \exp\left(-\frac{C_{14}}{T}\right) \]  
\[ (22) \]

\[ H(T) = C_{15} \exp\left(\frac{C_{16}}{T}\right) \]  
\[ (23) \]

\[ R_s(T) = C_{17} \exp\left(-\frac{C_{18}}{T}\right) \]  
\[ (24) \]
APPENDIX B. SAMPLE KEYWORD USED FOR SELF-PIERCE RIVETING PROCESS IN LS-DYNA

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7 1.7700E-6 45000.000  0.350000  473.00000  0.520000
$#     c1        c2        c3        c4        c5        c6        c7        c8
15.4869002.0800E-25  3.471150 1278.8600  0.0070481.1100E-11  0.1132507.1300E-34
Sheets

*PART
Punch

$# title
Pad and Punch

*MAT_RIGID_TITLE
Pad and Punch

$# title
Pad

$# title
Rivet

*MAT_PIECEWISE_LINEAR_PLASTICITY_TITLE
Steel

*HOURGLASS_TITLE
rivet

$# title

$# title
Usheet

<table>
<thead>
<tr>
<th>pid</th>
<th>secid</th>
<th>mid</th>
<th>eosid</th>
<th>hgid</th>
<th>grav</th>
<th>adpopt</th>
<th>tmid</th>
</tr>
</thead>
<tbody>
<tr>
<td>6</td>
<td>2</td>
<td>6</td>
<td>0</td>
<td>2</td>
<td>0</td>
<td>0</td>
<td>0</td>
</tr>
</tbody>
</table>

*MAT_BAMMAN_DAMAGE>Title

Mg damage top sheet

<table>
<thead>
<tr>
<th>mid</th>
<th>ro</th>
<th>e</th>
<th>pr</th>
<th>t</th>
<th>hc</th>
</tr>
</thead>
<tbody>
<tr>
<td>6</td>
<td>1.7700E-6</td>
<td>45000.000</td>
<td>0.3500000</td>
<td>473.000000</td>
<td>0.5200000</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>c1</th>
<th>c2</th>
<th>c3</th>
<th>c4</th>
<th>c5</th>
<th>c6</th>
<th>c7</th>
<th>c8</th>
</tr>
</thead>
<tbody>
<tr>
<td>15.4869002.0800E-25</td>
<td>3.471150</td>
<td>1278.8600</td>
<td>0.0070481.1100E-11</td>
<td>0.1132507.1300E-34</td>
<td></td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

<table>
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<tr>
<th>c9</th>
<th>c10</th>
<th>c11</th>
<th>c12</th>
<th>c13</th>
<th>c14</th>
<th>c15</th>
<th>c16</th>
</tr>
</thead>
<tbody>
<tr>
<td>1.910020</td>
<td>2085.4099</td>
<td>0.0005.7100E-132.1200E-335.2000E-19</td>
<td>2.400000</td>
<td>54.000000</td>
<td></td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>c17</th>
<th>a1</th>
<th>a2</th>
<th>a3</th>
<th>a4</th>
<th>a5</th>
<th>a6</th>
</tr>
</thead>
<tbody>
<tr>
<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
</tr>
</tbody>
</table>

*PART

$# title

Bottom compression

<table>
<thead>
<tr>
<th>pid</th>
<th>secid</th>
<th>mid</th>
<th>eosid</th>
<th>hgid</th>
<th>grav</th>
<th>adpopt</th>
<th>tmid</th>
</tr>
</thead>
<tbody>
<tr>
<td>7</td>
<td>2</td>
<td>8</td>
<td>0</td>
<td>2</td>
<td>0</td>
<td>0</td>
<td>0</td>
</tr>
</tbody>
</table>

*MAT_BAMMAN_DAMAGE>Title

Mg damage Compression

<table>
<thead>
<tr>
<th>mid</th>
<th>ro</th>
<th>e</th>
<th>pr</th>
<th>t</th>
<th>hc</th>
</tr>
</thead>
<tbody>
<tr>
<td>8</td>
<td>1.7700E-6</td>
<td>45000.000</td>
<td>0.3500000</td>
<td>473.000000</td>
<td>0.5200000</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>c1</th>
<th>c2</th>
<th>c3</th>
<th>c4</th>
<th>c5</th>
<th>c6</th>
<th>c7</th>
<th>c8</th>
</tr>
</thead>
<tbody>
<tr>
<td>15.4869002.0800E-25</td>
<td>3.471150</td>
<td>1278.8600</td>
<td>0.0070481.1100E-11</td>
<td>0.1132507.1300E-34</td>
<td></td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>c9</th>
<th>c10</th>
<th>c11</th>
<th>c12</th>
<th>c13</th>
<th>c14</th>
<th>c15</th>
<th>c16</th>
</tr>
</thead>
<tbody>
<tr>
<td>1.910020</td>
<td>2085.4099</td>
<td>0.0005.7100E-132.1200E-335.2000E-19</td>
<td>2.400000</td>
<td>54.000000</td>
<td></td>
<td></td>
<td></td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>c17</th>
<th>c18</th>
<th>a1</th>
<th>a2</th>
<th>a3</th>
<th>a4</th>
<th>a5</th>
<th>a6</th>
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</thead>
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<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
<td>0.000</td>
</tr>
</tbody>
</table>

*DEFINE_CURVE>Title

Displacement punch

<table>
<thead>
<tr>
<th>lcid</th>
<th>sidr</th>
<th>sfa</th>
<th>sfo</th>
<th>offa</th>
<th>offo</th>
<th>dattyp</th>
</tr>
</thead>
<tbody>
<tr>
<td>1</td>
<td>0</td>
<td>1.000000</td>
<td>1.000000</td>
<td>0.000</td>
<td>0.000</td>
<td>0</td>
</tr>
</tbody>
</table>

<table>
<thead>
<tr>
<th>a1</th>
<th>o1</th>
</tr>
</thead>
<tbody>
<tr>
<td>0.000</td>
<td>0.000</td>
</tr>
<tr>
<td>0.0500000</td>
<td>0.2500000</td>
</tr>
</tbody>
</table>
0.1068180   6.3000002
0.2000000   6.3000002

*DEFINE_CURVE_TITLE
Force
$#    lcid  sidr  sfa  sfo  offa  offo  dattyp
  2   0  1.000000  1.000000  0.000  0.000   0
$#
  0.000   4.000000
  0.4000000   4.0000000

*SET_PART_LIST_TITLE
SpringBack
$#    sid  da1  da2  da3  da4  solver
  7   0.000   0.000   0.000   0.000MECH
$#
  2   5   6   7   0   0   0   0

*CONSTRAINED_GLOBAL
$#    tc  rc  dir  x  y  z
  6   7   1   0.000   0.000   0.000

*INTERFACE_SPRINGBACK_LSDYNA
$#    psid  nshv  ftype
  7   35   2
$#
  7218   0   0

*ELEMENT_SHELL
*NODE
*END
APPENDIX C. SAMPLE KEYWORD USED FOR SPRINGBACK PROCESS IN LS-DYNA

*KEYWORD
*TITLE
$# title
LS-DYNA keyword deck by LS-PrePost
*CONTROL_BULK_VISCOSITY
1.500000 0.060000 1 0
*CONTROL>Contact
0.300000 0.000 1 2 4 1 1 0
0 0 0 0 4.000000 0 0 0
0.000 0.000 0.000 0.000 0.000 0.000 0.000
0 0 0 0 0 0 0 0.000
0 0 0 0.000 1.000000 0 0.000 0
0
*CONTROL_DYNAMIC_RELAXATION
$# nrcyck drtol dfctr drterm tssfdr irelal edttl idrflg
250 1.0000E-4 0.995000 0.000 0.000 0 0.040000 0 0.040000 -1
*CONTROL_ENERGY
1 2 1 1
*CONTROL_HOURGLASS
6 0.150000
*CONTROL_SHELL
20.000000 0 -1 1 15 2 2 0
1.000000 0 0 1 0 0 0 0
0 0 0 0 0 2
*CONTROL_TERMINATION
$# endtim endcyc dtmin endeng endmas
0.000 0.000 0.000 0.000 0.000
*CONTROL_TIMESTEP
0.000 0.900000 0 0.000 0.000 0 0 0
0.000 0 0
*DATABASE_BINARY_D3DRLF
$# dt lcmdt beam npltc psetid
1.000000 0 0 0 0
*DATABASE_BINARY_D3PLOT
5.0000E-4 0 0 0 0
0
*DATABASE_EXTENT_BINARY
35 35 3 1 1 1 1 1
0 0 0 1 1 1 2 1
0 0 1.000000 0 0 0 STRESS STRESS
<table>
<thead>
<tr>
<th>Command</th>
<th>Parameters</th>
</tr>
</thead>
<tbody>
<tr>
<td>*BOUNDARY_PRESCRIBED_MOTION_RIGID_ID</td>
<td>1Punch 3 2 2 1 -1.000000 0.000000E+28 0.000</td>
</tr>
<tr>
<td>*BOUNDARY_PRESCRIBED_MOTION_RIGID_ID</td>
<td>2Pad 4 2 2 1 -1.000000 0 0.050000 0.000</td>
</tr>
<tr>
<td>*LOAD_RIGID_BODY</td>
<td>4 2 2-150.00000 0 0 0 0</td>
</tr>
<tr>
<td>*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID</td>
<td>1Die Lsheet 2 1 1.000000 50 0.600000 0.500000 0.000 22</td>
</tr>
<tr>
<td></td>
<td>0.0001.00000E+20 1.000000 1.000000 0 0 0 0 0</td>
</tr>
<tr>
<td>*SET_PART_LIST_TITLE</td>
<td>Lsheet 2 0.000 0.000 0.000 0.000MECH</td>
</tr>
<tr>
<td></td>
<td>2 0 0 0 0 0 0 0 0</td>
</tr>
<tr>
<td>*SET_PART_LIST_TITLE</td>
<td>Die 1 0.000 0.000 0.000 0.000MECH</td>
</tr>
<tr>
<td></td>
<td>1 0 0 0 0 0 0 0 0</td>
</tr>
<tr>
<td>*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID</td>
<td>2Usheet Lsheet 2 6 1.000000 50 0.700000 0.450000 0.000 22</td>
</tr>
<tr>
<td></td>
<td>0.0001.00000E+20 1.000000 1.000000 0 0 0 0 0</td>
</tr>
<tr>
<td>*SET_PART_LIST_TITLE</td>
<td>Usheet 6 0.000 0.000 0.000 0.000MECH</td>
</tr>
<tr>
<td></td>
<td>6 0 0 0 0 0 0 0 0</td>
</tr>
<tr>
<td>*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID</td>
<td>3Pad Usheet 6 4 1.000000 50 0.600000 0.500000 0.000 22</td>
</tr>
<tr>
<td></td>
<td>0.0001.00000E+20 1.000000 1.000000 0 0 0 0 0</td>
</tr>
<tr>
<td>*SET_PART_LIST_TITLE</td>
<td>Pad 4 0.000 0.000 0.000 0.000MECH</td>
</tr>
<tr>
<td></td>
<td>4 0 0 0 0 0 0 0 0</td>
</tr>
<tr>
<td>*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID</td>
<td>4Rivet Usheet 6 5 1.000000 50 0.150000 0.120000 0.000 22</td>
</tr>
<tr>
<td></td>
<td>0.0001.00000E+20 1.000000 1.000000 0 0 0 0 0</td>
</tr>
<tr>
<td>*SET_PART_LIST_TITLE</td>
<td>Rivet 5 0.000 0.000 0.000 0.000MECH</td>
</tr>
<tr>
<td></td>
<td>5 0 0 0 0 0 0 0 0</td>
</tr>
<tr>
<td>*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID</td>
<td>5Rivet Lsheet 2 5 1.000000 50 0.150000 0.120000 0.000 22</td>
</tr>
<tr>
<td></td>
<td>0.0001.00000E+20 1.000000 1.000000 0 0 0 0 0</td>
</tr>
<tr>
<td>*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID</td>
<td></td>
</tr>
</tbody>
</table>
Punch Rivet

0.0001.0000E+20 1.000000 1.000000 0 0 0 0 0

*SET_PART_LIST_TITLE
Punch

*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
7Usheet Usheet

0.0001.0000E+20 1.000000 1.000000 0 0 0 0 0

*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
8Lsheet Lsheet

0.0001.0000E+20 1.000000 1.000000 0 0 0 0 0

*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
9Lsheet compression Lsheet

0.0001.0000E+20 1.000000 1.000000 0 0 0 0 0

*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
10Usheet compression Usheet

*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
11Lsheet compression Lsheet compression

0.0001.0000E+20 1.000000 1.000000 0 0 0 0 0

*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
12Die Lsheet compression

Die

*PART

Shell rigid

*MAT_RIGID_TITLE

Die

1.78500E-6 2.10000E+5 0.300000 0.000 0.000 0.000
1.000000 7 7
0.000 0.000 0.000 0.000 0.000 0.000
*PART
Lsheet
2 2 7 0 2 0 0 0
*SECTION_SHELL_TITLE
Shell non rigid
2 15 0.830000 6 1 0 0 1
0.100000 0.100000 0.100000 0.100000 0.000 0.000 0.000 0.000
*MAT_BAMMAN_DAMAGE_TITLE
Mg damage bottom tension
7 1.7700E-6 45000.000 0.350000 473.00000 0.520000
15.4869002.0800E-25 3.471150 1278.8600 0.0070481.1100E-11 0.1132507.1300E-34
1.910020 2085.4099 0.0057100E-132.1200E-335.2000E-19 2.400000 54.000000
0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000
0.268000 0.005000 8.000000
*HOURGLASS_TITLE
Sheets
2 6 0.100000 1 1.500000 0.060000 0.100000 0.100000
*PART
Punch
3 1 2 0 0 0 0 0
*MAT_RIGID_TITLE
Pad and Punch
2 7.8500E-6 2.10000E+5 0.300000 0.000 0.000 0.000
1.000000 6 7
0.000 0.000 0.000 0.000 0.000 0.000
*PART
Pad
4 1 2 0 0 0 0 0
*PART
Rivet
5 2 4 0 1 0 0 0
*MAT PIECEWISE_LINEAR_PLASTICITY_TITLE
Steel
4 7.8500E-6 2.10000E+5 0.300000 880.000000 300.000000 0.000 0.000
0.000 0.000 0 0 0.000
0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000
0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000
*HOURGLASS_TITLE
rivet
1 6 0.150000 1 1.500000 0.060000 0.150000 0.150000
*PART
Usheet
6 2 6 0 2 0 0 0
*MAT_BAMMAN_DAMAGE_TITLE
Mg damage top sheet
6 1.7700E-6 45000.000 0.350000 473.00000 0.520000
15.4869002.0800E-25 3.471150 1278.8600 0.0070481.1100E-11 0.1132507.1300E-34

41
PART
Bottom compression
 7 2 8 0 2 0 0 0

*MAT_BAMMAN_DAMAGE
Mg damage Compression
 8 1.77E6 45000.00 0.350000 473.0000 0.520000
15.48E6 902.08E-25 3.471150 1278.8600 0.0070481.1100E-11 0.1132507.1300E-34
1.91E02 2085.4099 0.005710E-132.1200E-335.2000E-19 2.400000 54.000000
0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000
0.800000 0.0005000 15.000000

*MAT_BAMMAN
Bammann
 5 1.77E6 45000.00 0.350000 473.0000 0.520000
28.67E991.9500E-12 3.421150 1278.8600 2.634870 1.2910E-31
3.58E272 1744.8956 1.0400E-258.8600E-114.0700E-115.2700E-164.5530E-17 0.02054
0.000 0.000 0.000 0.000 0.000 0.000 0.000 0.000

*DEFINE_CURVE
Displacement punch
 1 0 1.000000 1.000000 0.000 0.000 0
0.000 0.000
0.050000 0.250000
0.1068180 6.3000002
0.2000000 6.3000002

*DEFINE_CURVE
Force
 2 0 1.000000 1.000000 0.000 0.000 0
0.000 4.0000000
0.4000000 4.0000000

*DEFINE_CURVE
DR
$# lcid sidr sfa sfo offa offo dattyp
 3 1 1.000000 1.000000 0.000 0.000 0
$#
a1 o1
0.000 0.000
1.0000000 1.0000000

*SET_PART_LIST
SpringBack
 7 0.000 0.000 0.000 0.000MECH
2 5 6 7 0 0 0 0

*CONSTRAINED
2 6 7 1 0.000 0.000 0.000

*INTERFACE
Springback
 7 35 2
7218 0 0

*NODE

*ELEMENT_SHELL_THICKNESS
*INITIAL_STRESS_SHELL
*INITIAL_STRAIN_SHELL
*END
APPENDIX D. SAMPLE KEYWORD USED FOR COOLING PROCESS IN LS-DYNA

*KEYWORD
*TITLE
$# title
LS-DYNA keyword deck by LS-PrePost

*CONTROL_BULK_VISCOSITY
$# q1 q2 type btype
  1.500000  0.060000   1  0

*CONTROL_CONTACT
$# slsfac rwpanl ischk shthk penopt thkchg orien enmass
  0.300000   0.000  1  2  4  1  1  0
$# usrfac usrfrc nsbcs interm xpene ssthk edct tiedprj
  0   0   0   4.000000   0   0   0
$# sfric dfrc edc vfc th th_sf pen_sf
  0.000  0.000  0.000  0.000  0.000  0.000  0.000
$# ignore fceng skiprwg outseg spotstp spotdel spothin
  0   0   0   0   0   0   4.000000   0   0.000
$# isym nserod rwgaps rwgdth rwksf icov swradf ithoff
  0   0   0   0.000   1.000000   0   0.000   0
$# shledg

*CONTROL_DYNAMIC_RELAXATION
$# nrcky drtol dfcrtr drterm tssfdt irelal edttl idrlflg
  250 1.0000E-4  0.995000   0.000   0.000   0.000  0.040000  -1

*CONTROL_ENERGY
$# hgen rwen slnten ryylen
  1   2   1   1

*CONTROL_HOURGLASS
$# ihq qh
  6 0.150000

*CONTROL_SHELL
$# wrpang esort irnxx istupd theory wbc miter proj
  20.000000   0 -1  1   15   2   2   0
$# rotascl intgrd lamshl cstyp6 tshell nfail1 nfail4 pfnfail
  1.000000   0   1   0   0   0   0
$# psst upd irquad cntco itsflg irquad
  0   0   0   0   0   2

*CONTROL_TERMINATION
$# endtim endcyc dtmin endeng endmas
  0.000   0   0.000   0.000   0.000

*CONTROL_TIMESTEP
### DATABASE_BINARY_D3DRLF
```plaintext
*DATABASE_BINARY_D3DRLF
```

### DATABASE_BINARY_D3PLOT
```plaintext
*DATABASE_BINARY_D3PLOT
```

### DATABASE_EXTENT_BINARY
```plaintext
*DATABASE_EXTENT_BINARY
```

### BOUNDARY_PRESCRIBED_MOTION_RIGID_ID
```plaintext
*BOUNDARY_PRESCRIBED_MOTION_RIGID_ID
```

### LOAD_RIGID_BODY
```plaintext
*LOAD_RIGID_BODY
```

### CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
```plaintext
*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
```

### SET_PART_LIST_TITLE
```plaintext
*SET_PART_LIST_TITLE
```

### Lsheet
```plaintext
Lsheet
```

```plaintext
Die Lsheet
```

```plaintext
*BOUNDARY_PRESCRIBED_MOTION_RIGID_ID
```

```plaintext
*LOAD_RIGID_BODY
```

```plaintext
*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
```

```plaintext
*SET_PART_LIST_TITLE
```

```plaintext
Die
```

```plaintext
*SET_PART_LIST_TITLE
```

*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID

$#  cid  title
   2Usheet Lsheet
$#  sids  sidm sfact  freq  fs  fd  dc  membs
       2  6  1.000000  50  0.700000  0.450000  0.000  22
$#  tbirth  tdeath  sos  som  nds  ndm  cof  init
       0.0001  1.0000E+20  1.000000  1.000000  0  0  0  0

*SET_PART_LIST_TITLE
Usheet
$#  sid  da1  da2  da3  da4  solver
       6  0.000  0.000  0.000  0.000  MECH
$#  pid1  pid2  pid3  pid4  pid5  pid6  pid7  pid8
       6  0  0  0  0  0  0  0

*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID

$#  cid  title
   3Pad Usheet
$#  sids  sidm sfact  freq  fs  fd  dc  membs
       6  4  1.000000  50  0.600000  0.500000  0.000  22
$#  tbirth  tdeath  sos  som  nds  ndm  cof  init
       0.0001  1.0000E+20  1.000000  1.000000  0  0  0  0

*SET_PART_LIST_TITLE
Pad
$#  sid  da1  da2  da3  da4  solver
       4  0.000  0.000  0.000  0.000  MECH
$#  pid1  pid2  pid3  pid4  pid5  pid6  pid7  pid8
       4  0  0  0  0  0  0  0

*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID

$#  cid  title
   4Rivet Usheet
$#  sids  sidm sfact  freq  fs  fd  dc  membs
       6  5  1.000000  50  0.150000  0.120000  0.000  22
$#  tbirth  tdeath  sos  som  nds  ndm  cof  init
       0.0001  1.0000E+20  1.000000  1.000000  0  0  0  0

*SET_PART_LIST_TITLE
Rivet
$#  sid  da1  da2  da3  da4  solver
       5  0.000  0.000  0.000  0.000  MECH
$#  pid1  pid2  pid3  pid4  pid5  pid6  pid7  pid8
       5  0  0  0  0  0  0  0

*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID

$#  cid  title
   5Rivet Lsheet
$#  sids  sidm sfact  freq  fs  fd  dc  membs
       2  5  1.000000  50  0.150000  0.120000  0.000  22
$#  tbirth  tdeath  sos  som  nds  ndm  cof  init
       0.0001  1.0000E+20  1.000000  1.000000  0  0  0  0
47

```plaintext
$# cid title
6Punch Rivet
$# sids sidm sfact freq fs fd dc membs
 5  3  1.000000  50  0.740000  0.570000  0.000  22
$# tbirth tdeath sos som nds ndm cof init
 0.0001.0000E+20  1.000000  1.000000  0  0  0  0  0
*SET_PART_LIST_TITLE
Punch
$# sid da1 da2 da3 da4 solver
 3  0.000  0.000  0.000  0.000MECH
$# pid1 pid2 pid3 pid4 pid5 pid6 pid7 pid8
 3  0  0  0  0  0  0  0
*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
$# cid title
7Usheet Usheet
$# sids sidm sfact freq fs fd dc membs
 6  6  1.000000  50  0.700000  0.450000  0.000  22
$# tbirth tdeath sos som nds ndm cof init
 0.0001.0000E+20  1.000000  1.000000  0  0  0  0  0
*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
$# cid title
8Lsheet Lsheet
$# sids sidm sfact freq fs fd dc membs
 2  2  1.000000  50  0.700000  0.450000  0.000  22
$# tbirth tdeath sos som nds ndm cof init
 0.0001.0000E+20  1.000000  1.000000  0  0  0  0  0
*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
$# cid title
9Lsheet compression Lsheet
$# sids sidm sfact freq fs fd dc membs
 2  8  1.000000  50  0.700000  0.450000  0.000  22
$# tbirth tdeath sos som nds ndm cof init
 0.0001.0000E+20  1.000000  1.000000  0  0  0  0  0
*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
$# cid title
bottom Compression
$# sid da1 da2 da3 da4 solver
 8  0.000  0.000  0.000  0.000MECH
$# pid1 pid2 pid3 pid4 pid5 pid6 pid7 pid8
 7  0  0  0  0  0  0  0
*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
$# cid title
10Lsheet compression Usheet
$# sids sidm sfact freq fs fd dc membs
 6  8  1.000000  50  0.700000  0.450000  0.000  22
$# tbirth tdeath sos som nds ndm cof init
 0.0001.0000E+20  1.000000  1.000000  0  0  0  0  0
*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID
$# cid title
11Lsheet compression Lsheet compression
```
$#  sids  sidm  sfact  freq  fs  fd  dc  membs  
8   8  1.000000  50  0.700000  0.450000  0.000  22
$#  tbirth  tdeath  sos  som  nds  ndm  cof  init  
0.0001.0000E+20  1.000000  1.000000  0  0  0  0  0  
*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID  
$#  cid  
12 Die Lsheet compression  
$#  sids  sidm  sfact  freq  fs  fd  dc  membs  
8   1  1.000000  50  0.600000  0.500000  0.000  22
$#  tbirth  tdeath  sos  som  nds  ndm  cof  init  
0.0001.0000E+20  1.000000  1.000000  0  0  0  0  0
*CONTACT_2D_AUTOMATIC_SURFACE_TO_SURFACE_ID  
$#  cid  
13 Rivet Lsheet compression  
$#  sids  sidm  sfact  freq  fs  fd  dc  membs  
8   5  1.000000  50  0.150000  0.120000  0.000  22
$#  tbirth  tdeath  sos  som  nds  ndm  cof  init  
0.0001.0000E+20  1.000000  1.000000  0  0  0  0  0
*PART  
$#  title  
Die
$#  pid  secid  mid  eosid  hgid  grav  adpopt  tmid  
1   1   1   0   0   0   0   0
*SECTION_SHELL>Title  
Shell rigid
$#  secid  elform  shrf  nip  propt  qr/irid  icomp  setyp  
1  15  0.830000  4  1  0  0  1
$#  t1  t2  t3  t4  nloc  marea  idof  edgset  
0.100000  0.100000  0.100000  0.100000  0.000  0.000  0.000  0
*MAT_RIGID_TITLE  
Die
$#  mid  ro  e  pr  n  couple  m  alias  
1  7.85000E-6  2.10000E+5  0.300000  0.000  0.000  0.000
$#  cmo  con1  con2  
1.000000  7  7
$#  lco  or  a1  a2  a3  v1  v2  v3  
0.000  0.000  0.000  0.000  0.000  0.000
*MAT_BAMMAN_DAMAGE_TITLE  
Lsheet
$#  pid  secid  mid  eosid  hgid  grav  adpopt  tmid  
2   2   7   0   2   0   0   0
*SECTION_SHELL>Title  
Shell non rigid
$#  secid  elform  shrf  nip  propt  qr/irid  icomp  setyp  
2  15  0.830000  6  1  0  0  1
$#  t1  t2  t3  t4  nloc  marea  idof  edgset  
0.100000  0.100000  0.100000  0.100000  0.000  0.000  0.000  0
**Mg damage bottom tension**

```
$#     mid  ro  e  pr  t  hc
    7  1.7700E-6  45000.000  0.350000  473.00000  0.520000
$#   c1   c2   c3   c4   c5   c6   c7   c8
  15.4869002.0800E-25  3.471150  1278.8600  0.0070481.1100E-11  0.1132507.1300E-34
$#   c9   c10  c11  c12  c13  c14  c15  c16
  1.910020  2085.4099  0.0057100E-132.1200E-335.2000E-19  2.400000  54.000000
$#   c17  c18  a1   a2   a3   a4   a5   a6
       0.000  0.000  0.000  0.000  0.000  0.000  0.000  0.000
$#   n   d0   fs
  0.268000  0.005000  8.000000
```

*HOURGLASS_TITLE*

**Sheets**

```
$# hgid   ihq   qm   ibq   q1   q2   qb/vdc   qw
  2     6  0.100000  1  1.500000  0.060000  0.100000  0.100000
```

*PART*

**Punch**

```
$#  pid   secid   mid   eosid   hgid   grav   adpopt   tmid
  3  1  2  0  0  0  0
```

*MAT_RIGID_TITLE*

**Pad and Punch**

```
$#  mid   ro   e   pr   n   couple   m   alias
  2  7.8500E-6  2.1000E+5  0.300000  0.000  0.000  0.000
$#  cmo   con1   con2
  1.000000  6  7
$# lco or a1   a2   a3   v1   v2   v3
       0.000  0.000  0.000  0.000  0.000  0.000
```

*PART*

**Pad**

```
$#  pid   secid   mid   eosid   hgid   grav   adpopt   tmid
  4  1  2  0  0  0  0
```

*PART*

**Rivet**

```
$#  pid   secid   mid   eosid   hgid   grav   adpopt   tmid
  5  2  4  0  1  0  0  0
```

*MAT_PIECEWISE_LINEAR_PLASTICITY_TITLE*

**Steel**

```
$#  mid   ro   e   pr   sigy   etan   fail   tdel
  4  7.8500E-6  2.1000E+5  0.300000  880.00000  300.00000  0.000  0.000
$#  c   p   lcss   lcsr   vp
       0.000  0.000  0  0  0.000
$#  eps1   eps2   eps3   eps4   eps5   eps6   eps7   eps8
       0.000  0.000  0.000  0.000  0.000  0.000  0.000  0.000
$#  es1   es2   es3   es4   es5   es6   es7   es8
       0.000  0.000  0.000  0.000  0.000  0.000  0.000  0.000
```

*HOURGLASS_TITLE*
rivet
$#  hgid  ihq  qm  ibq  q1  q2  qb/vdc  qw
  1   6   0.150000  1  1.500000  0.060000  0.150000  0.150000
*PART
$# title
Usheet
$#  pid  secid  mid  eosid  hgid  grav  adpopt  tmid
   6   2   6   0   2   0   0   0
*MAT_BAMMAN_DAMAGE_TITLE
Mg damage top sheet
$#  mid  ro  e  pr  t  hc
   6  1.7700E-6  45000.000  0.350000  473.000000  0.520000
$#  c1  c2  c3  c4  c5  c6  c7  c8
15.4869002.0800E-25  3.471150  1278.8600  0.0070481.1100E-11  0.1132507.1300E-34
$#  c9  c10  c11  c12  c13  c14  c15  c16
1.910020  2085.4099  0.0005.7100E-132.1200E-335.2000E-19  2.400000  54.000000
$#  c17  c18  a1  a2  a3  a4  a5  a6
0.000  0.000  0.000  0.000  0.000  0.000  0.000  0.000
$#  n  d0  fs
0.268000  0.005000  4.000000
*PART
$# title
Bottom compression
$#  pid  secid  mid  eosid  hgid  grav  adpopt  tmid
   7   2   8   0   2   0   0   0
*MAT_BAMMAN_DAMAGE_TITLE
Mg damage Compression
$#  mid  ro  e  pr  t  hc
   8  1.7700E-6  45000.000  0.350000  473.000000  0.520000
$#  c1  c2  c3  c4  c5  c6  c7  c8
15.4869002.0800E-25  3.471150  1278.8600  0.0070481.1100E-11  0.1132507.1300E-34
$#  c9  c10  c11  c12  c13  c14  c15  c16
1.910020  2085.4099  0.0005.7100E-132.1200E-335.2000E-19  2.400000  54.000000
$#  c17  c18  a1  a2  a3  a4  a5  a6
0.000  0.000  0.000  0.000  0.000  0.000  0.000  0.000
$#  n  d0  fs
0.800000  0.005000  15.000000
*MAT_BAMMAN_TITLE
Bammann
$#  mid  ro  e  pr  t  hc
   5  1.7700E-6  45000.000  0.350000  473.000000  0.520000
$#  c1  c2  c3  c4  c5  c6  c7  c8
28.6770991.9500E-12  3.421150  1278.8600  2.634870  1.2910E-4  0.0397138.0400E-31
$#  c9  c10  c11  c12  c13  c14  c15  c16
3.582272  1744.89561.0400E-198.8600E-114.0700E-115.2700E-164.5530E-17  0.002054
$#  c17  c18  a1  a2  a3  a4  a5  a6
0.000  0.000  0.000  0.000  0.000  0.000  0.000  0.000
*INITIAL_STRAIN_SHELL
*INITIAL_STRESS_SHELL
*DEFINE_CURVE_TITLE
Displacement punch
$#  lcid  sidr  sfa  sfo  offa  offo  dattyp
  1   0  1.000000  1.000000  0.000  0.000   0
$#
     a1     o1
  0.000   0.000
  0.050000  0.2500000
  0.1068180   6.3000002
  0.2000000   6.3000002
*DEFINE_CURVE_TITLE
Force
$#  lcid  sidr  sfa  sfo  offa  offo  dattyp
  2   0  1.000000  1.000000  0.000  0.000   0
$#
     a1     o1
  0.000   4.0000000
  0.4000000   4.0000000
*DEFINE_CURVE_TITLE
DR
$#  lcid  sidr  sfa  sfo  offa  offo  dattyp
  3   1  1.000000  1.000000  0.000  0.000   0
$#
     a1     o1
  0.000   0.000
  1.0000000   1.0000000
*SET_PART_LIST_TITLE
SpringBack
$#  sid  da1  da2  da3  da4  solver
  7   0.000  0.000  0.000  0.000  MECH
$#
    pid1  pid2  pid3  pid4  pid5  pid6  pid7  pid8
  2    5    6    7    0    0    0    0
*CONSTRAINED_GLOBAL
$#  tc  rc  dir  x  y  z
  6    7   1  0.000  0.000  0.000
*INTERFACE_SPRINGBACK_LSDYNA
$#  psid  nshv  ftype
  7    35    2
$#
    nid  tc  rc
  7218   0   0
*ELEMENT_SHELL_THICKNESS
*NODE
*END